

# Factors Affecting Qualification/Certification:

## The Effect of Machine Variabilities on Mechanical Properties of LB-PBF Ti-6Al

-4V Gr. 5

National Center for Additive Manufacturing Excellence (NCAME)



JAMS Technical Review  
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# Introduction

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Project Title: Factors Affecting Qualification/Certification The Effect of Machine Variabilities on Mechanical Properties of Additive Manufactured Materials

Principal Investigator Nima Shamsaei

(See next slide for complete list of participants.)

FAA Technical Monitor: Kevin Stonaker

Source of matching contribution: Faculty time and use of equipment



# Project team

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## Advisory Group



### NCAME Project Team Auburn University

**PI:** Nima Shamsaei (Mechanical Engr.)

**PM:**

**Co-PIs:**

Shuai Shao (Mechanical Engr.)

Masoud Mahjouri Samani (Elec. Comp. Engr.)

Hareesh Tippur (Mechanical Engr.)



# Background

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Effect of key process variables (KPVs) drift within the tolerances on defect content and part performance is not very well understood

Identified on several Roadmaps

Challenge arises from the dependency of microdefect-structure and mechanical properties to multiple synergistic factors, including powder quality, laser-material interaction, inherent heat transfer effects, geometrical factors, process parameters etc.



# Challenge

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# Objective & approach

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Objective: To understand the effect of KPVs drift within tolerance bands on defect characteristics and mechanical properties of ~~PIBF~~ T6Al-4V Gr. 5

Approach: Four steps are taken,

- I. Quantify the powder reuse effects to exclude their influences from the KPVs drift study. Geometry and time interval will be kept constant
- II. Identify the combined effect of KPVs (laser power, laser spot size, and hatching distance) drift and location on the defect structure
- III. Evaluate the impact of KPVs drift on tensile, fatigue, and high strain rate fracture behaviors using specimens fabricated with worst KPVs/location combinations
- IV. Leverage machine learning and simulations wherever applicable





### TASK 1: Literature Review

Collect information from literature, equipment OEM, FAA AM team, and steering committee

### TASK 2: Design of Experiments

Design and refine the experiments for PBF T6Al-4V Gr. 5 based on findings from the literature review

### TASK 3: Powder Release Effects at $\Delta T = 17.71j (-0.595 (0.17) \mu m) \rightarrow 29 (6-143.5) \rightarrow 15.43 (2.964) \rightarrow 23.56$





DoE to understand the effect of powder reuse



# Powder reuse effect

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Prints start with unused powder and continues until powder is used several times

48 fatigue specimens (ASTM E466) (12 in each quadrant)

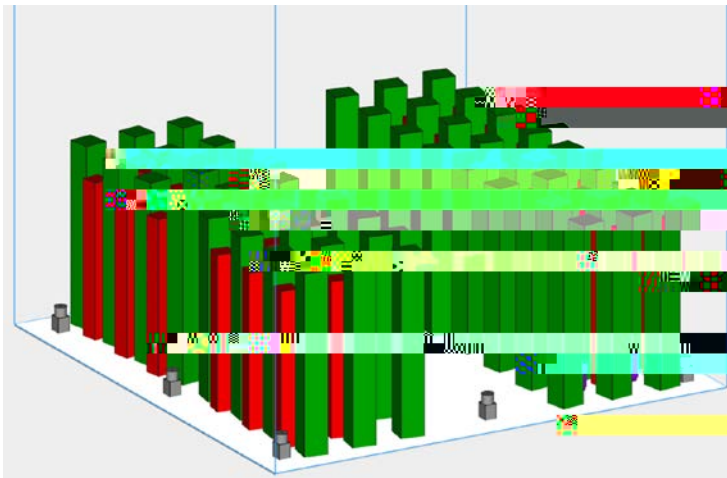
24 tensile specimens (ASTM E8) (6 in each quadrant)

16 high-strain test specimens (4 in each quadrant)

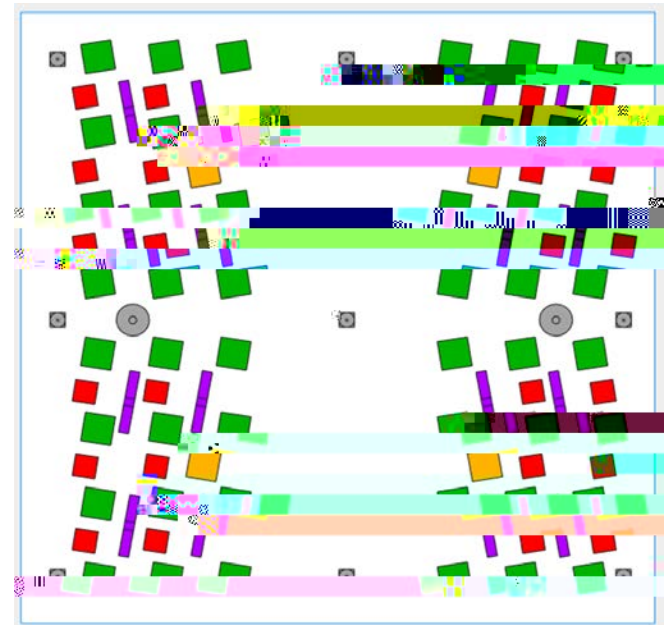
2 powder holders (at both ends) Note that the powder in E and W may be sampled based on ASTM B215 instead of using the powder holders

4 microscopy specimens (1 in each quadrant)

9 XCT witness coupons



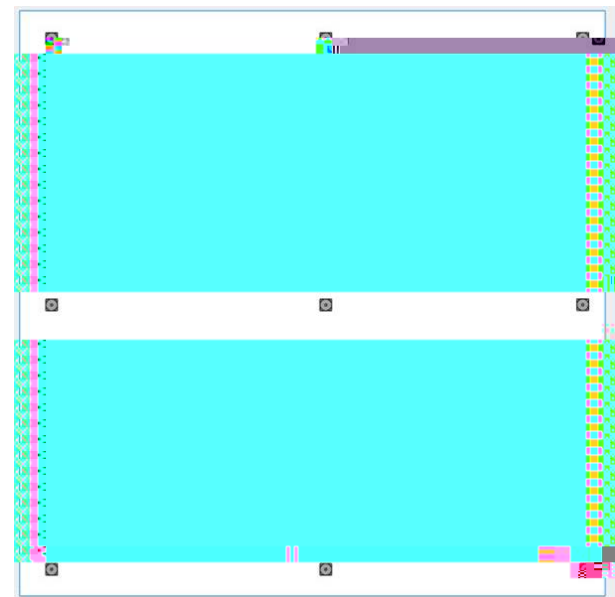
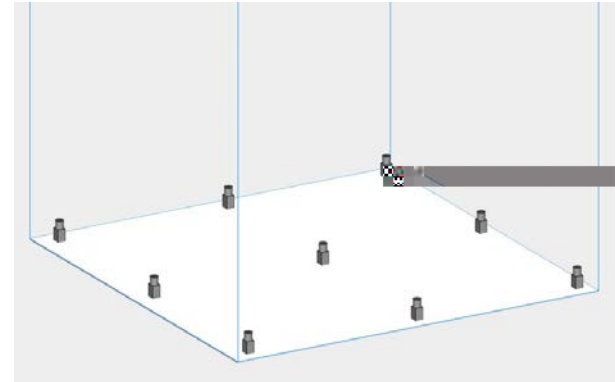
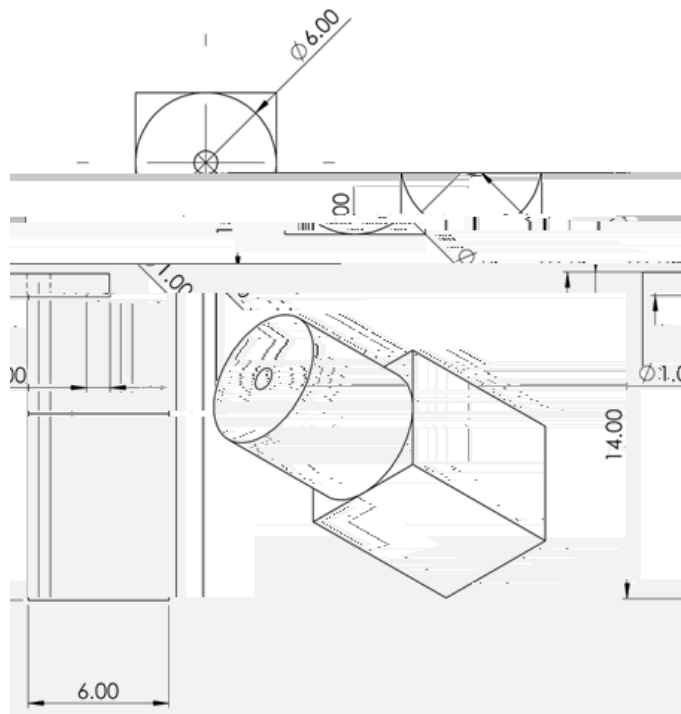
Occupied build plate density= 12%



# Build quality XCT coupons

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XCT porosity witness coupons are always included in all prints throughout the study



DoE to identify the worst combination of  
KPVs/location



# Identify worst KPVs combinations

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Suggested KPVs and their possible deviations from nominal values:

Laser power  $\pm 4\%$

Scan speed  $\pm 0.015\%$  (accuracy decreases for scan speeds  $> 3000$  mm/s)

Layer thickness  $\pm 0.1$   $\mu\text{m}$

Hatching distance  $\pm 2.4\%$

Laser spot size (focal diameter)  $80$   $\mu\text{m} \pm 5$   $\mu\text{m}$  (highly depends on material and laser powder, not monitored)

Scan speed and layer thickness are excluded from this study due to their significant drifts

Only laser power (P), laser spot size (d) and hatching distance (h) will be included for further analyses

P, d and h will be either individually or simultaneously altered to increase or decrease the energy density (E) from the baseline

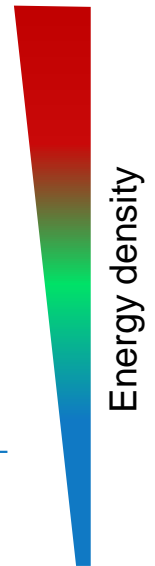


# Identify worst KPVs combinations

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Only **overheating** and **underheating** KPV combinations will be investigated as they can intensify the defect population

Factorial Design Parameters		Hatch spacing (h)		
Laser power (P)	Laser spot size (d)	$h^+$	$h^\circ$	$h^-$
$P^+$	$d^+$	$P^+ d^+ h^+$	$P^+ d^+ h^\circ$	$P^+ d^+ h^-$
	$d^\circ$	$P^+ d^\circ h^+$	<b><math>P^+ d^\circ h^\circ</math></b>	<b><math>P^+ d^\circ h^-</math></b>
	$d^-$	$P^+ d^- h^+$	<b><math>P^+ d^- h^\circ</math></b>	<b><math>P^+ d^- h^-</math></b>
$P^\circ$	$d^+$	$P^\circ d^+ h^+$	$P^\circ d^+ h^\circ$	<b><math>P^\circ d^+ h^-</math></b>
	$d^\circ$	$P^\circ d^\circ h^+$	<b><math>P^\circ d^\circ h^\circ</math></b>	<b><math>P^\circ d^\circ h^-</math></b>
	$d^h$			









DoE to determine impact of KPVs drift on mechanical performance



# Effect of KPVs drift on properties

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All specimens will be fabricated based on the worst combination of KPVs/location

XCT coupons will be fabricated via recommended KPVs to monitor the effect of powder reuse (whenever possible)

The tensile specimens will be machined to solely focus on the defects and microstructure of the parts

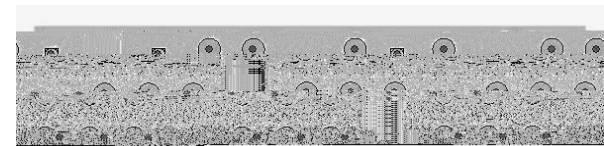
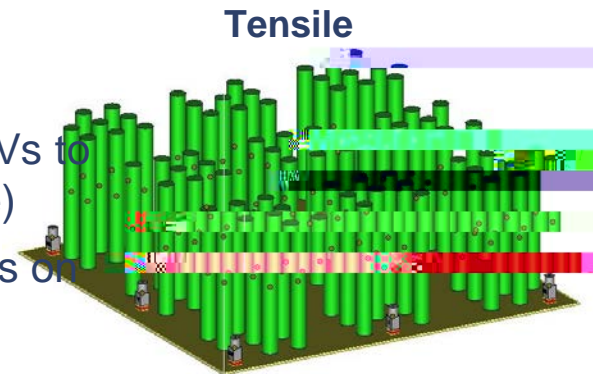
The fabrication will be repeated at least multiple times for the tensile specimens to have 8 specimens per condition

136 tensile specimens (17 sets of 8 specimens)

6 specimens will be tested for each condition and kept as backup

9 XCT coupons per built plate

\* The number of specimens may be adjusted to minimize the occupied build plate density to avoid excessive spatter, and powder layer disturbance due to part interaction



Occupied build plate density = 10%







# Preliminary results



# Powderre-use effect: defect characteristics

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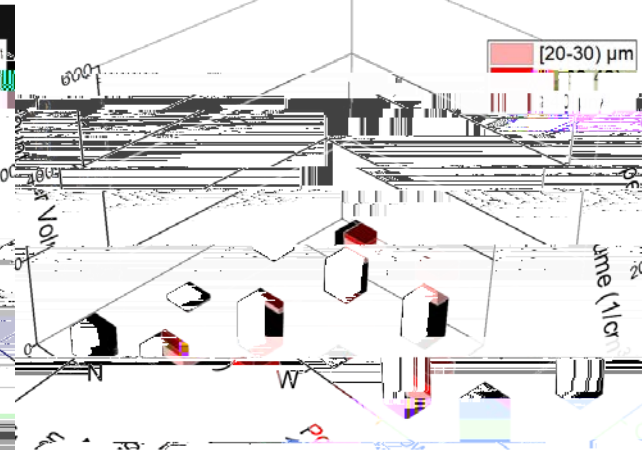
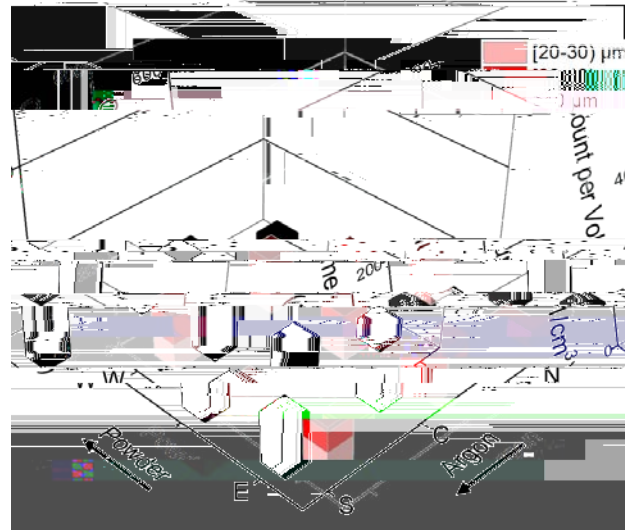
South locations tend to produce more defects

Defect population tend to decrease in used powder

Virgin Powder

3X Reused Powder

7X Reused Powder











# Location dependency and KPVs drift effects

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SE or SW generally performed worse for most KPVs.

A volume of  $\sim(5 \times 5 \times 5) \text{ mm}^3$  was considered for all specimens

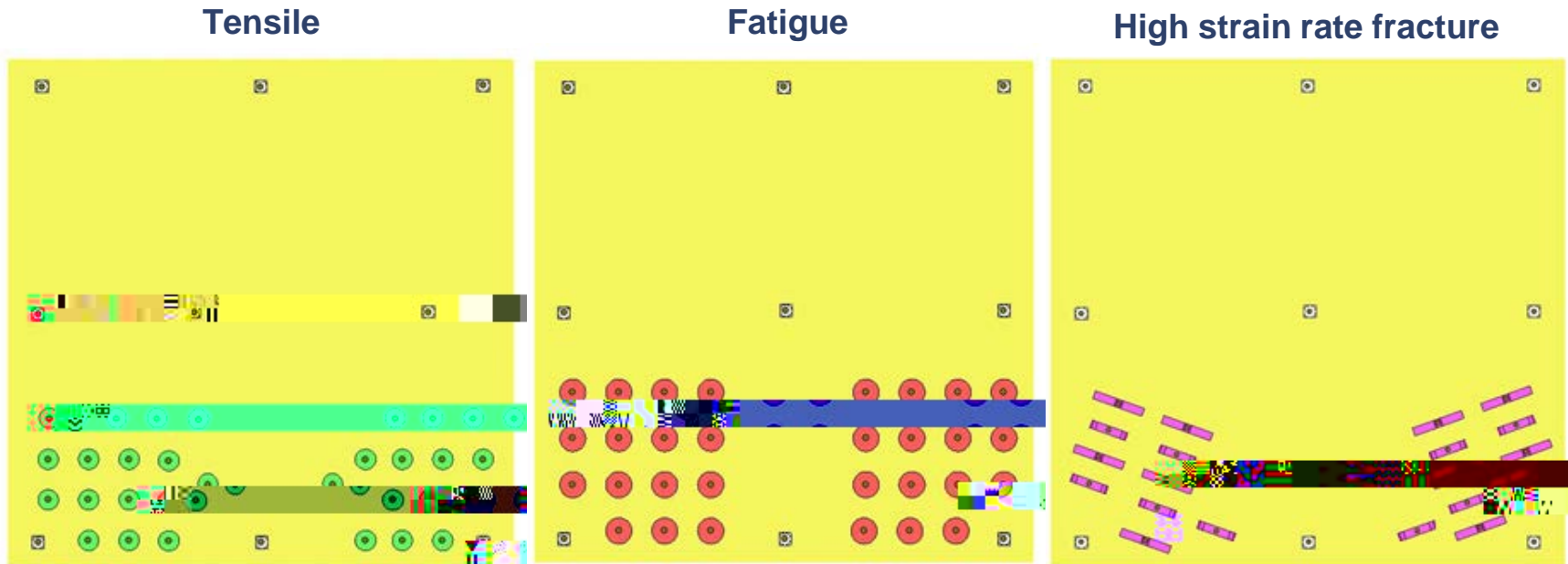
Gray box indicates overestimation of defect size due to the smas cyo(z)198.0.5 (iackgro5(d)



# Next step

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Fabrication of specimens will be limited to SW and SE locations as they were found to produce the worst defect conditions



Data on powder reuse is planned to be fully analyzed and become available by the end of 2021

